

1. (Currently Amended) An axial-flow thermal turbomachine, ~~having~~ comprising:  
a rotor (1) made from a metallic material with a first density ( $D_1$ ), in which;  
a circumferential groove; and  
rotor blades (3, 3') and intermediate pieces (4) are alternately mounted alternately in a  
the circumferential groove, characterized in that;  
wherein said intermediate pieces (4) consist of comprise a material with a second density  
( $D_2$ ), which is lower than the first density ( $D_1$ ).
2. (Currently Amended) The turbomachine as claimed in claim 1, ~~characterized in that wherein~~ the material having the second density ( $D_2$ ) ~~is~~ comprises an intermetallic compound.
3. (Currently Amended) The turbomachine as claimed in claim 2, ~~characterized in that the wherein~~ said intermetallic compound ~~is~~ comprises an alloy selected from the group  
consisting of a  $\gamma$ -titanium aluminide alloy or and an orthorhombic titanium aluminide alloy.
4. (Currently Amended) The turbomachine as claimed in claim 3, ~~characterized in that the wherein~~ said  $\gamma$ -titanium aluminide alloy has the following chemical composition (details in % by weight): Ti-(30.5-31.5)Al-(8.9-9.5)W-(0.3-0.4)Si.
5. (Currently Amended) The turbomachine as claimed in claim 1, ~~characterized in that wherein~~ the material having the second density ( $D_2$ ) ~~is~~ comprises a titanium alloy.
6. (Currently Amended) The turbomachine as claimed in ~~one of claims Claim 1 to 5,~~  
~~characterized in that wherein~~ the turbomachine ~~is~~ comprises a gas turbine having a high-pressure  
compressor of a gas turbine having with a rotor (1) which ~~substantially~~ comprises a stainless Cr-Ni steel.